



MERTIS – the design of a highly integrated IR imaging spectrometer

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Background

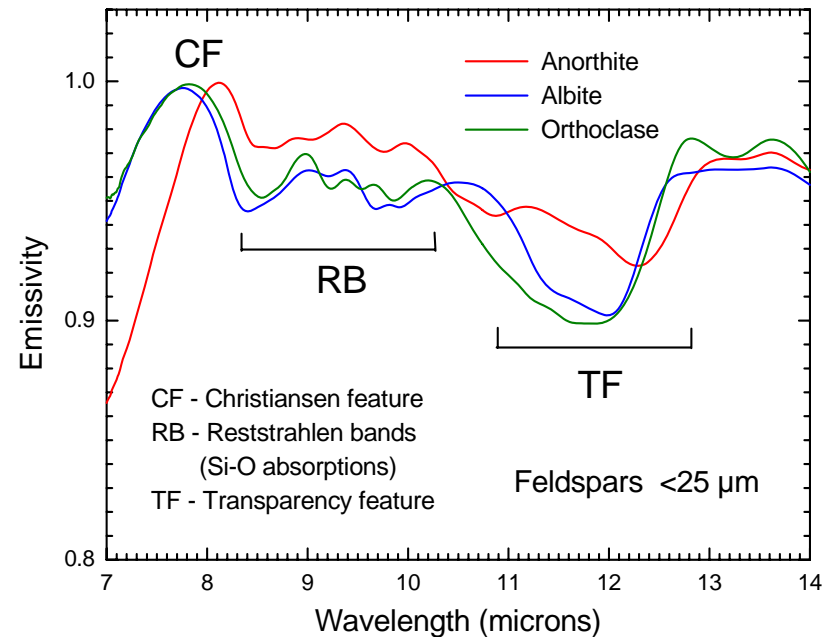
- Mercury has probably the oldest surface among the terrestrial planets with large daily temperature variations
- Investigation of the mineralogical composition and thermo-physical properties are motivating measurements of the spectral emittance and the radiometric behavior in the IR

Study of Mercury's surface composition
Identification of rock-forming minerals
Mapping of the surface mineralogy

Spectrometer objectives

Study of surface temperature and thermal inertia

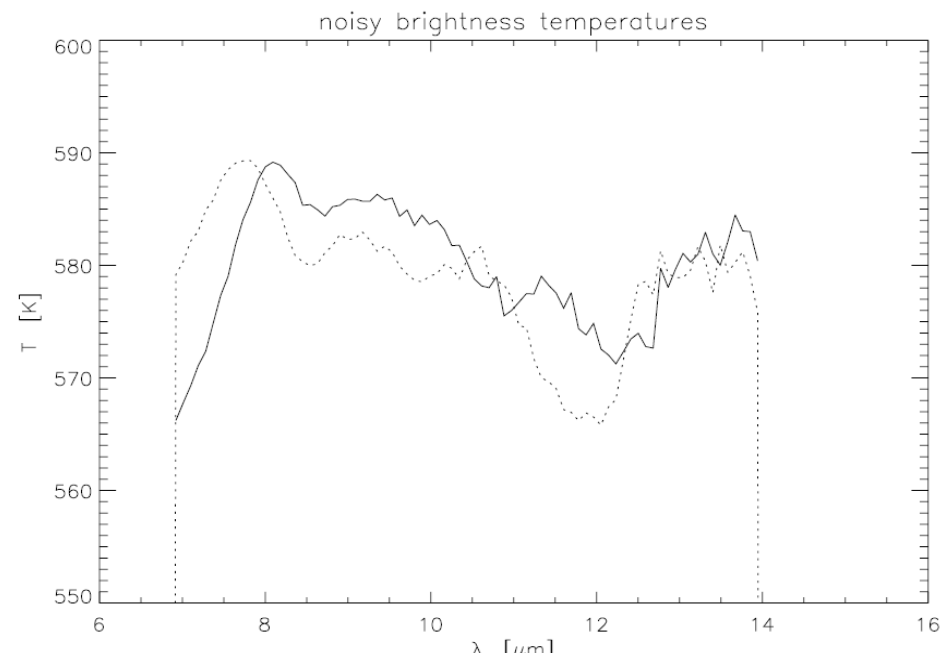
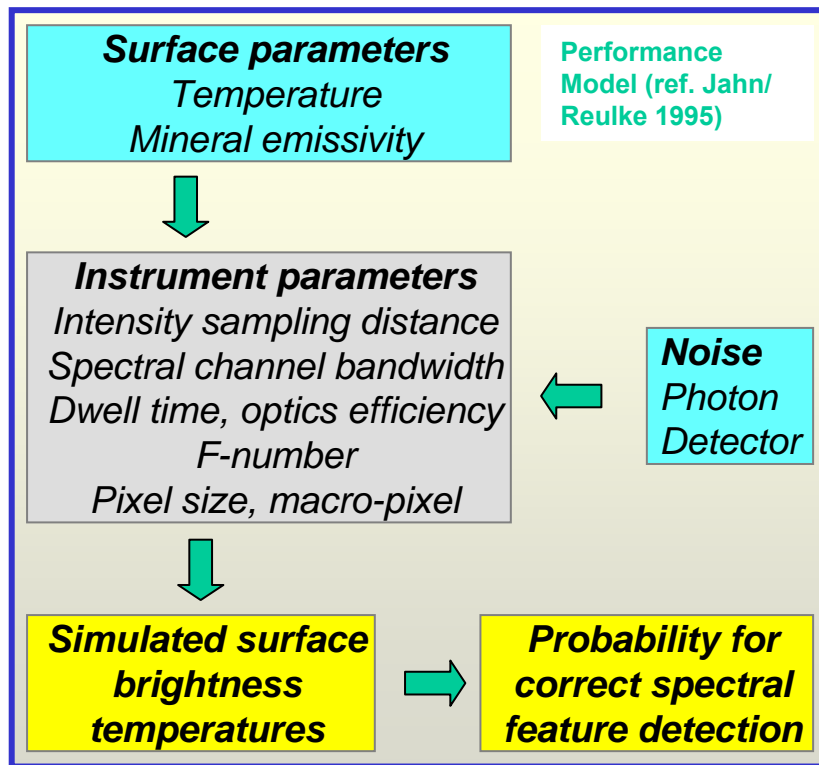
Radiometer objective



Laboratory emittance spectra of fine-grained feldspar (ref. Wagner 2000)

Instrument Simulations

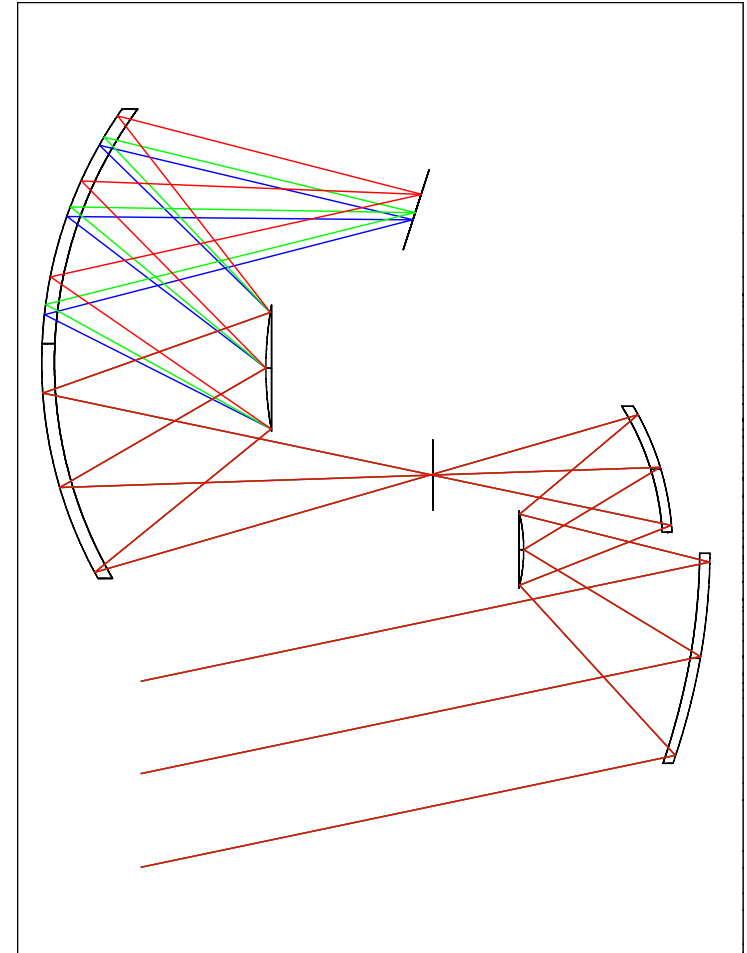
- Scientific performance assessment for initial instrument architecture
- Simulation of spectral signal and noise values depending on Mercury surface data and instrument parameters



as initial, 2x pixel size

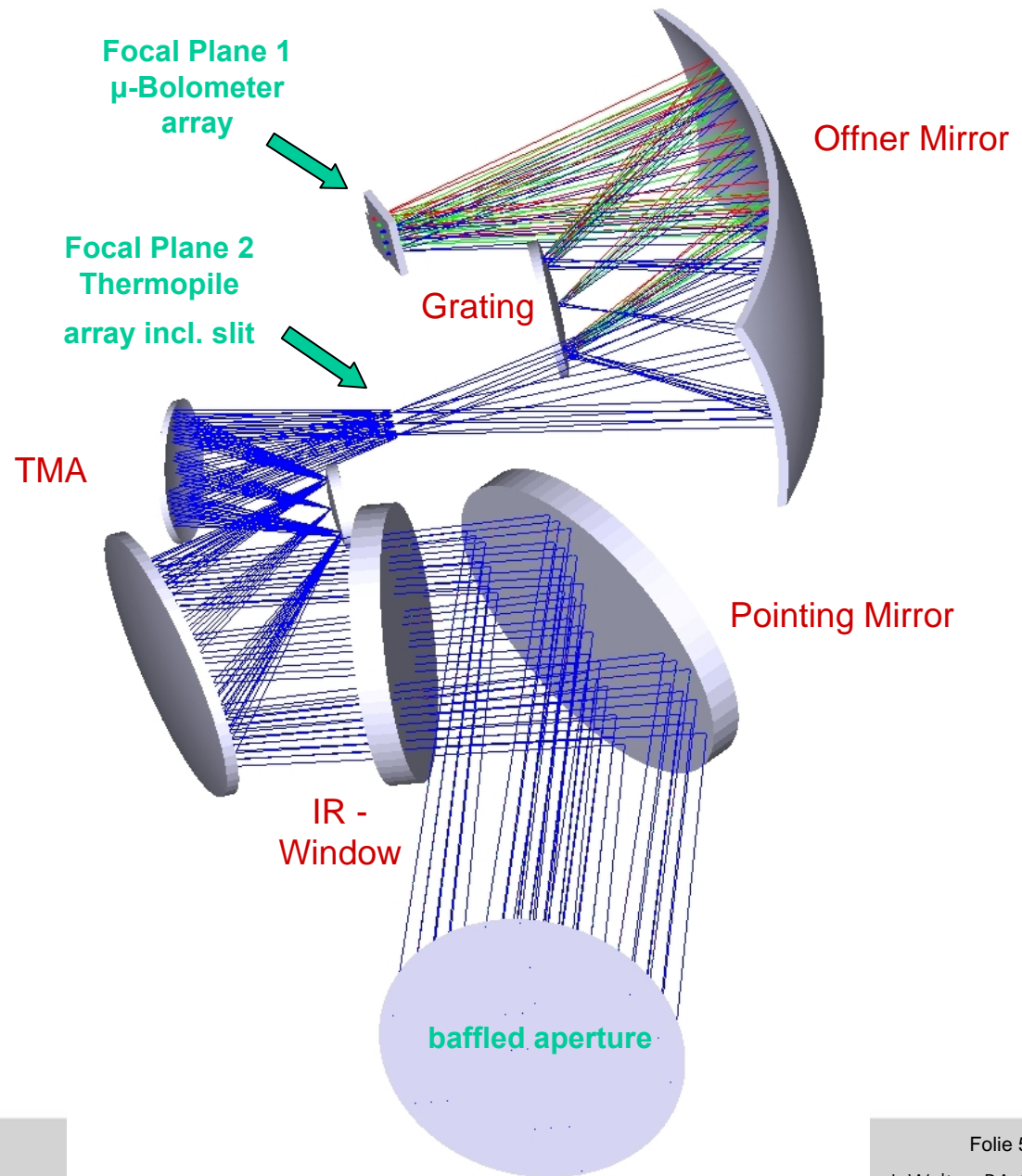
Instrument Concept

- o **MERTIS – MERcury Thermal Infrared Spectrometer**
- o IR-imaging spectrometer based on the push-broom principle (80 channels @ 90 nm)
- o Application of un-cooled micro-bolometer array providing spectral separation and spatial resolution according to its 2- dimensional shape
- o Operation concept principle is characterized by intermediate scanning of the planet surface and 3 different calibration targets – free space and on-board black body sources
- o In - field separated micro-radiometer based on thermopile line arrays

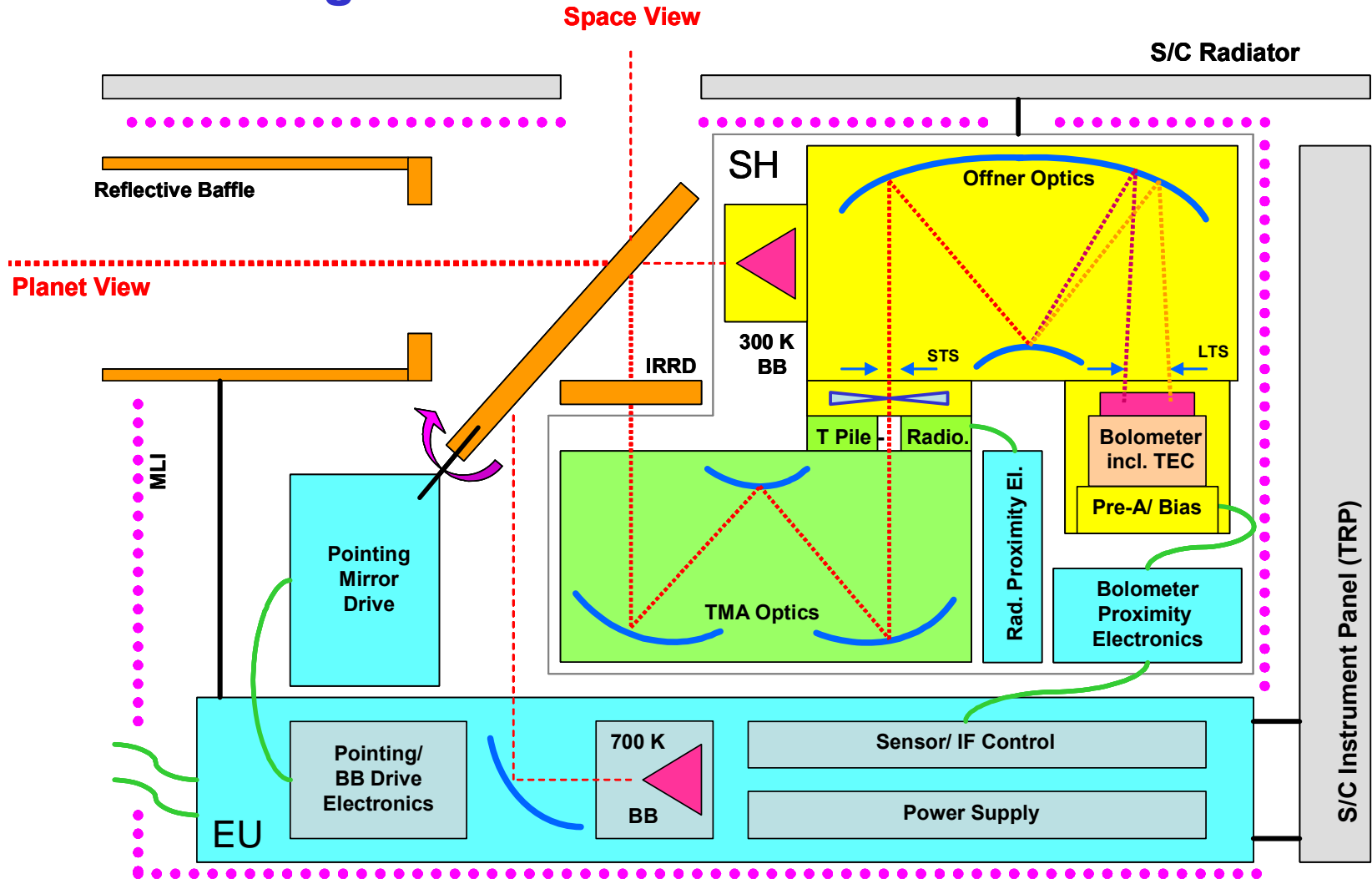


Optics

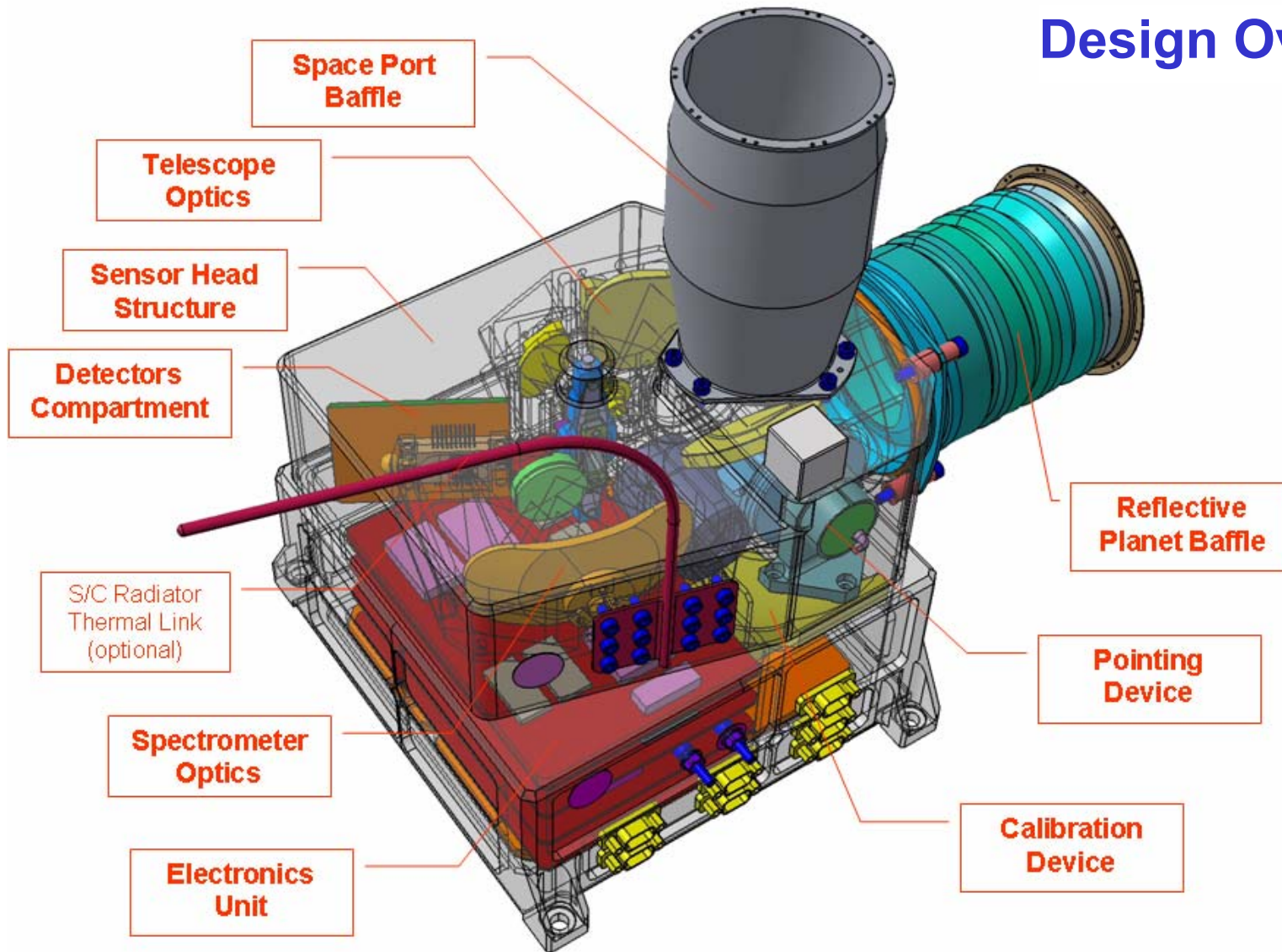
- o All - reflective optics design for high efficiency
(0.95 mirrors * 0.57 grating)
- o Off - axis TMA behind a IR- entrance window
(F# 2 / 50, 7- 40 μm transparency)
- o Offner spectrometer including convex grating
(blaze structure 90 x 5 μm , angle 3.7°)
- o Innovative integrated dual focal plane concept
- o Pointing mirror to target orientation
(FOV 4° x 1° each)



Block diagram



Design Overview

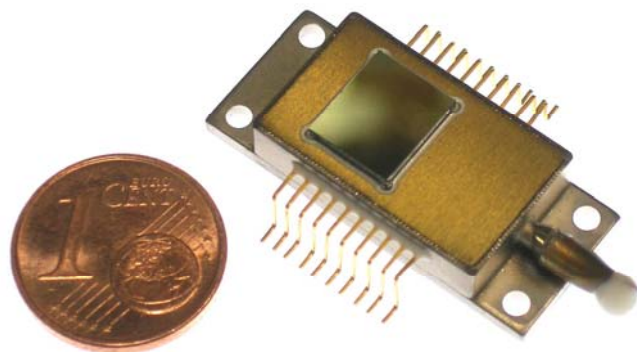
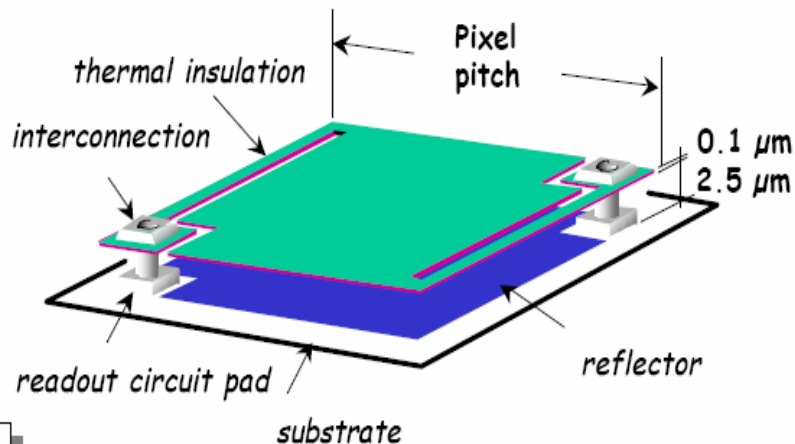
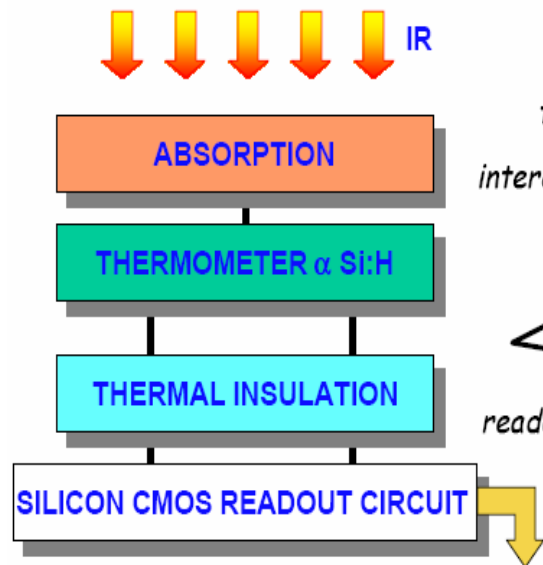


Main Parameters

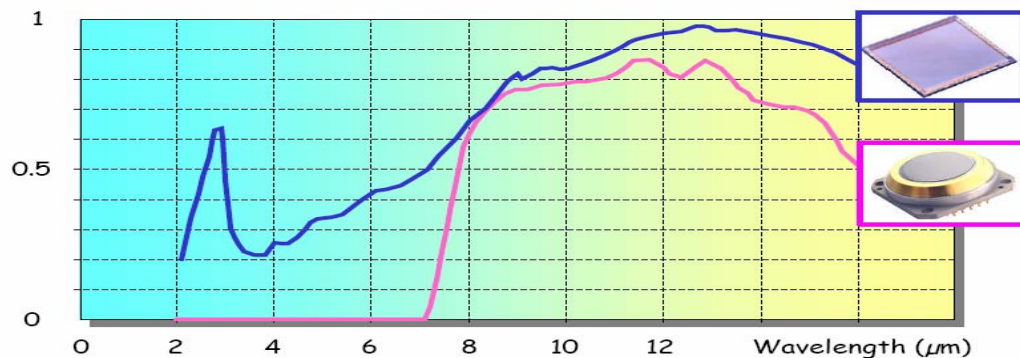
Parameter	Unit	Spectrometer	Radiometer (μRAD)
Focal length	F	50 mm	
F – number	F#	2.0	
Optical efficiency	η_{opt}	0.54	
Microbolometer array detector <ul style="list-style-type: none"> illuminated pixels 	pixels	160 x 120 @ 35 μm 100 spatial 80 spectral	
μRAD thermopile line array		2 x 15 @ 250 μm	
Spectral channel width	$\lambda\delta$	90 nm / pixel	
Spectral resolution	$\lambda/\lambda\delta$	78 – 156	
Spectral range	λ	7 – 14 μm	7 – 40 μm
Detectivity	D^*	$0.95 \cdot 10^9 \text{ cm Hz}^{1/2} \text{ W}^{-1}$	$7 \cdot 10^8 \text{ cm Hz}^{1/2} \text{ W}^{-1}$
Instantaneous field of view	IFOV	0.7 mrad	5 mrad
Ground sample distance <ul style="list-style-type: none"> Periherm 400 km Apoherm 1500 km 	GSD	280 - 1400 m (M = 1- 5) 1050 m	2000 m 7500 m
Dwell time <ul style="list-style-type: none"> Periherm 400 km Apoherm 1500 km 	τ	109 ms 784 ms	775 ms 5597 ms
Field of view	FOV	4° ACT, 0° ALT	4° ACT, 1° ALT
Swath width		28 km	
Instrument overall dimensions + ext. Baffle		140 x 160 x 120 mm ³ 93 x ø54 & 92 x ø68 mm ³	
Instrument total mass incl. 20% margin		2.85 kg	

Detector Spectrometer

- o Resistive semi-conductor micro-bolometer
(Array 120 x 160 pixels @ 35 μm pitch)
- o Thermally controlled MEMS Device
(TEC at 10 mK stability)
- o Established technology to be space-qualified for BepiColombo

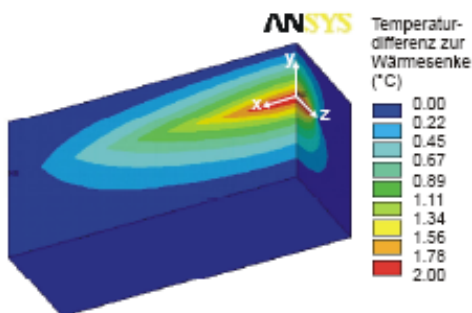


COTS detector for bread-boarding



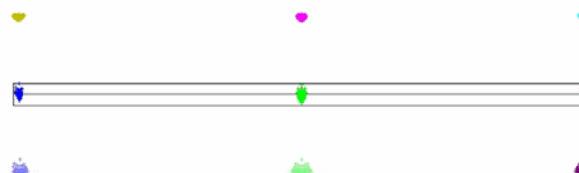
Detector Radiometer

- o Thin film IR thermopile with thermoelectric high-effective material
(2 line arrays 2 x 15 pixels @ 250 μm pitch)
- o Spaced in-parallel from the optical axis
(0,5 mm from slit centre)
- o Technology study and performance simulations ongoing

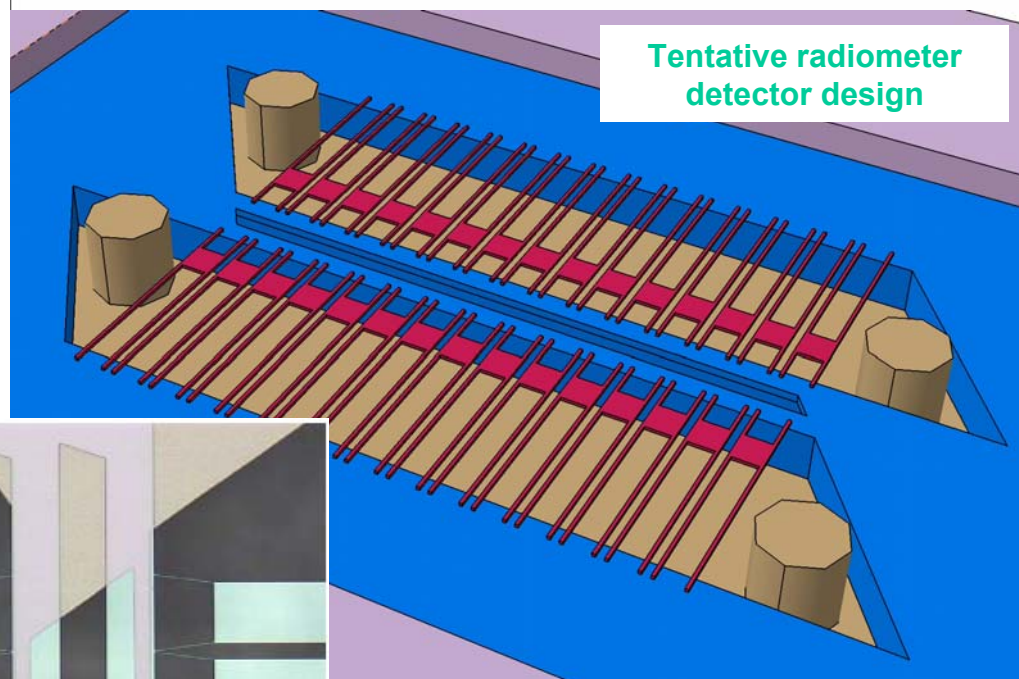


FEM simulation and experimental Si-wafer for tests

Optical design spot image at slit

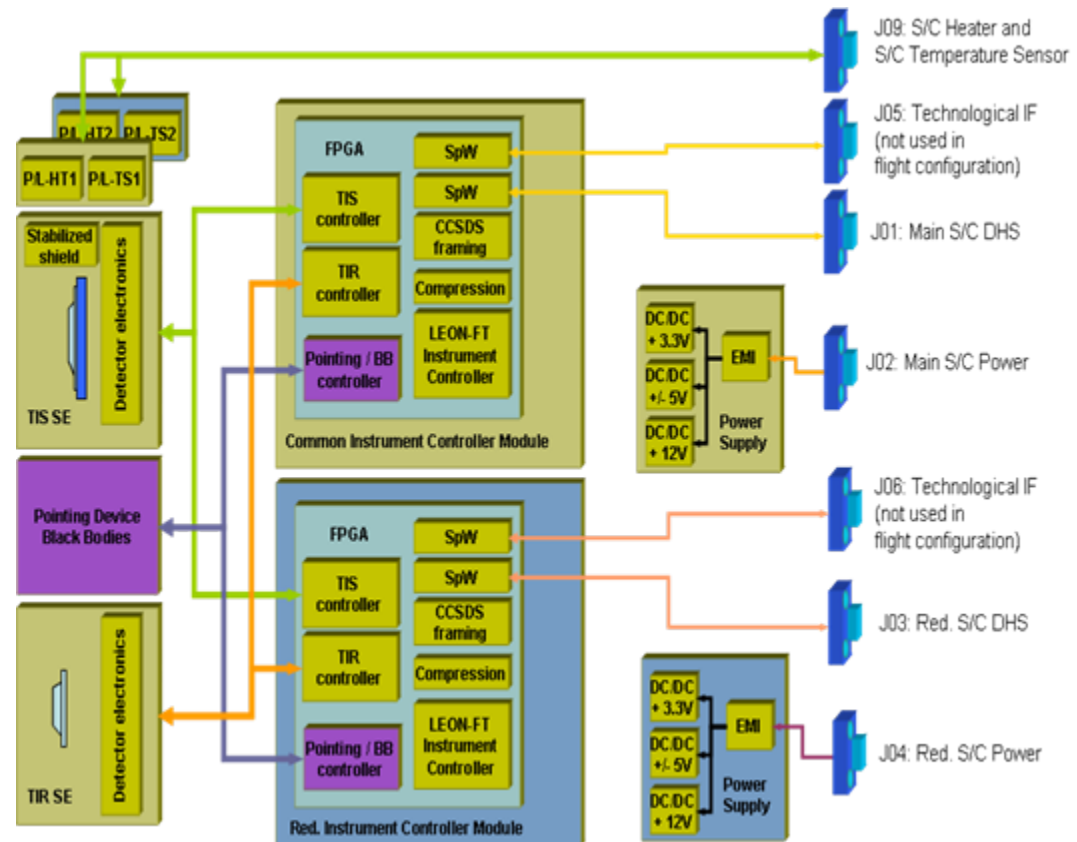


Tentative radiometer detector design



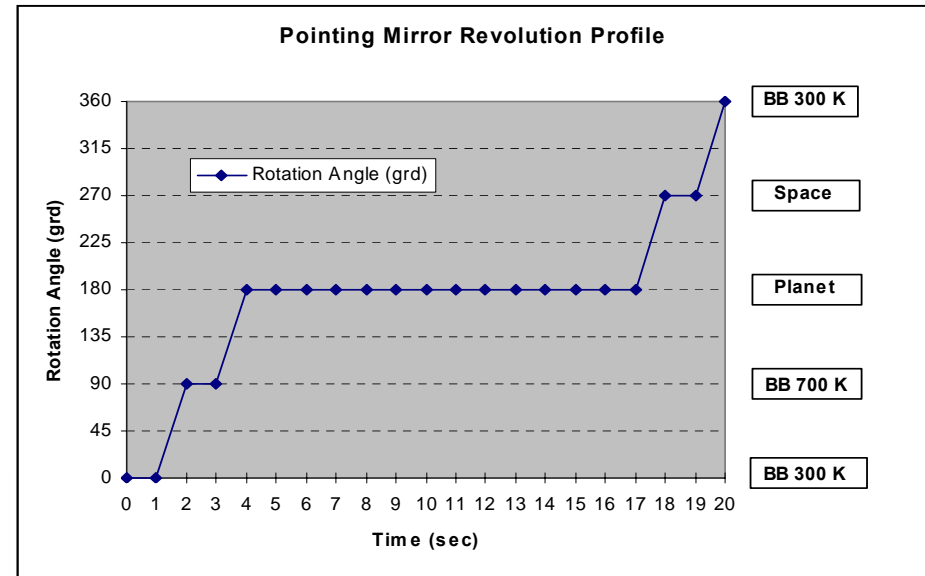
Electronics Architecture

- o Modular concept based on FPGA technology
- o Cold redundancy of main electronics parts
- o Tasks
 - Independent control of the MERTIS sub-systems
 - Acquisition and processing of science data
 - Providing of internal voltages and interfacing the +28V S/C power bus
 - TMC management and interfacing to the S/C DHU
 - Control of the pointing mirror black bodies and optional shutters
 - Providing of HK and status information
 - Control of the detector
 - Temperature stabilization of the detector
 - Detector signal conditioning



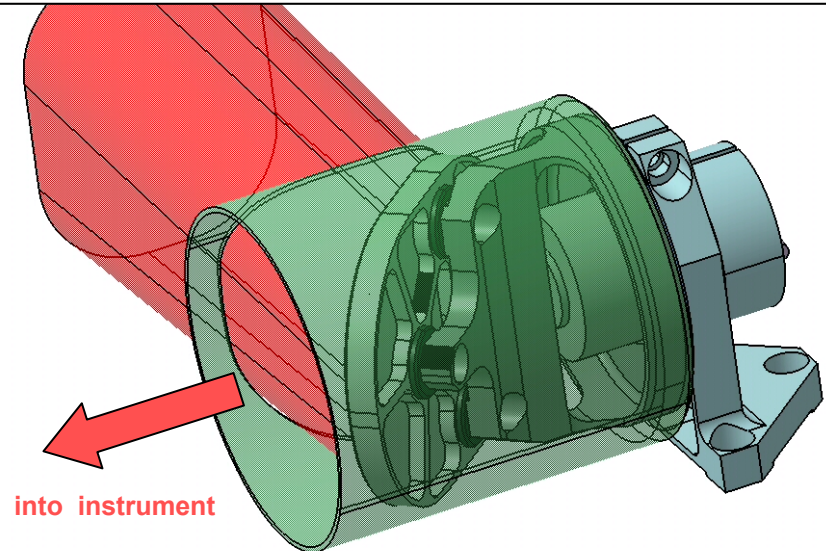
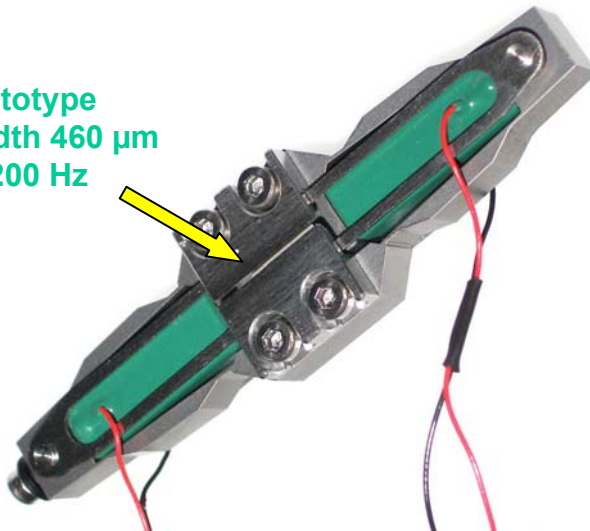
Development - Actuators

- o In-flight calibration purposes
- o Pointing Unit for target selection
 - Planet (65 % duty cycle min.)
 - Deep space
 - 300 K black body
 - 700 K black body
- o Short Term Shutter for instrument temperature reference
 - Operated every dwell time (~10 Hz)



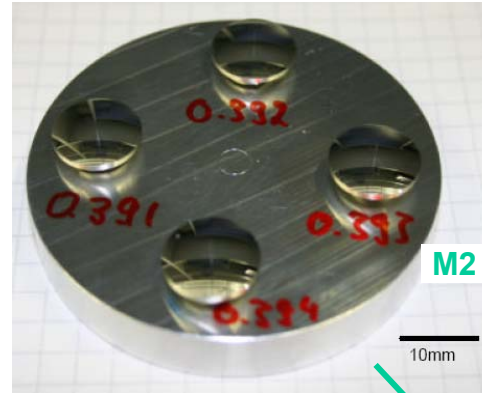
STS prototype

- Slit width 460 μm
- $F_{\text{res}} \sim 200 \text{ Hz}$

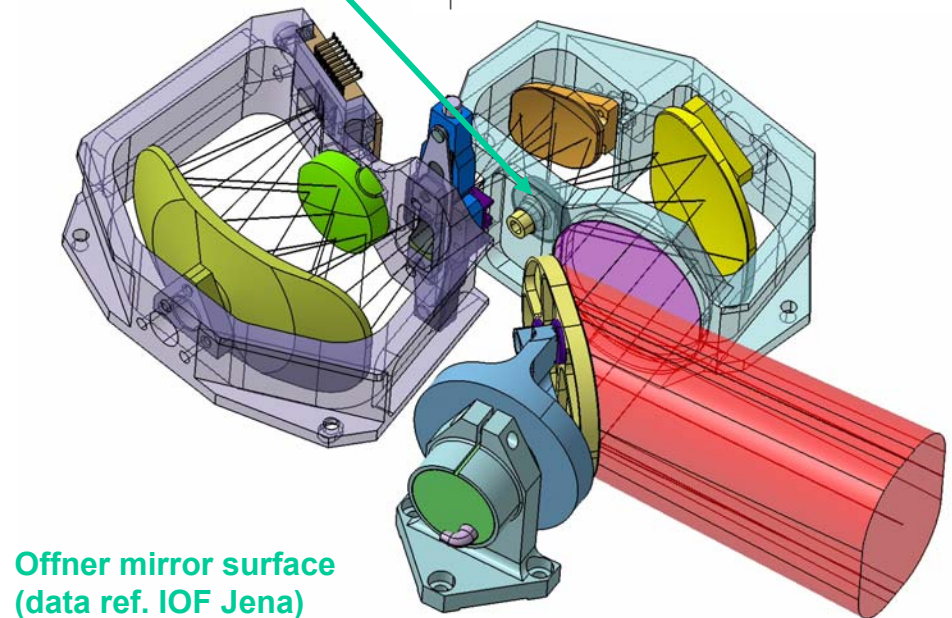
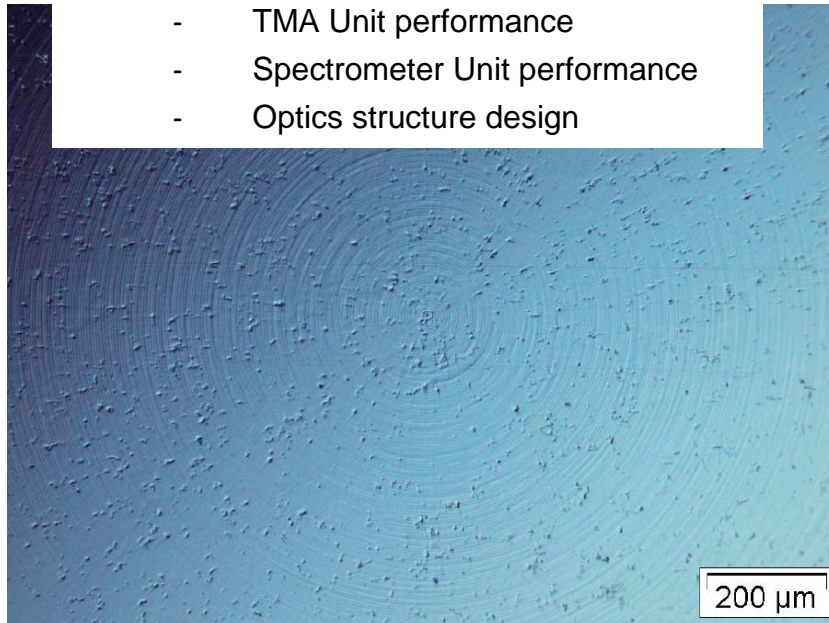
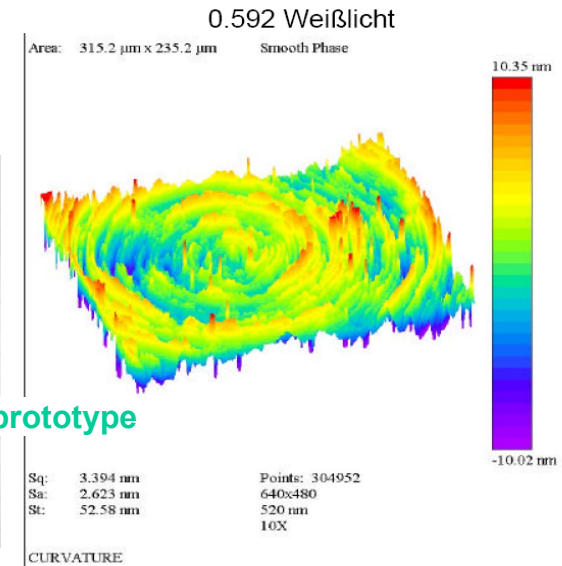


Development – Optics Technology

- o Early identification of flight optics technology
 - Single point diamond turning
 - All aluminium + gold coating
 - Element parameters and efficiency measurements (grating)
- o Verification of Phase A Sensor Head design
 - TMA Unit performance
 - Spectrometer Unit performance
 - Optics structure design



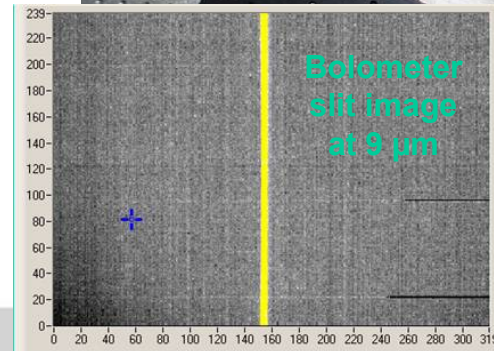
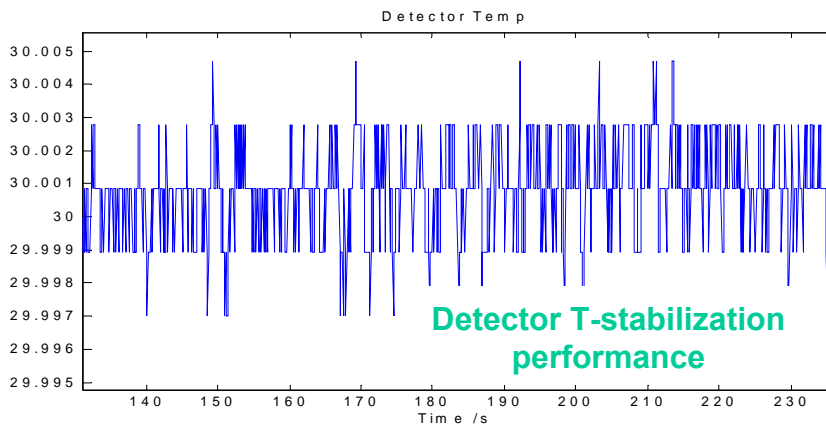
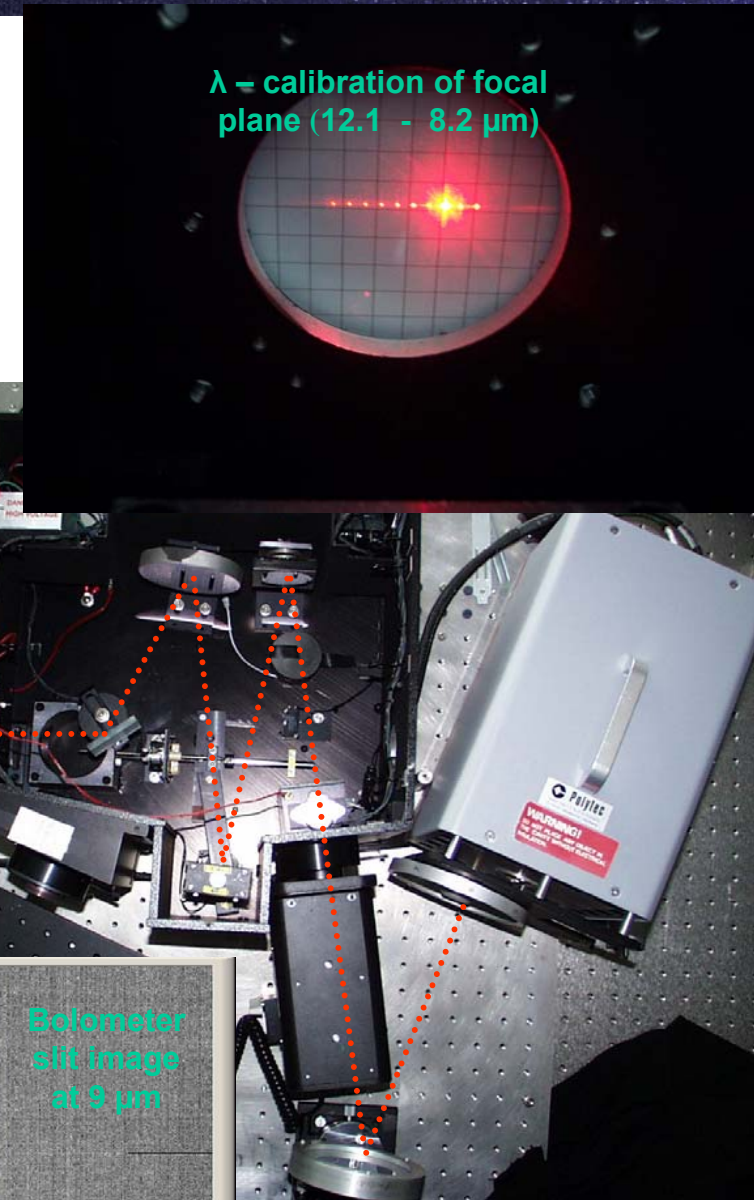
M2 prototype



Offner mirror surface
(data ref. IOF Jena)

Development – Laboratory Work

- o Radiometric Analysis Breadboard (RAB)
 - Investigations of the environment stability (lab conditions), pattern correction and calibration approach
 - Adaptation and verification of MERTIS components (detector, proximity electronics, EGSE)
- o Spectro Radiometric Breadboard (SRB)
 - Investigations of the optical performance of TMA and spectrometer (grating, detector)
 - Verification of the MERTIS baseline design (grating, detector, F#, D*)



See you in 2018 !



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